#### 2024

#### M.Sc.

# **Fourth Semester** DISCIPLINE SPECIFIC ELECTIVE – 04 **MATHEMATICS**

Course Code: MMAD 4.21 (Fluid Mechanics)

Total Mark: 70 Time: 3 hours Pass Mark: 28

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Answer five questions, taking one from each unit.

## UNIT-I

- 1. (a) Describe the Lagrange's and Eulerian methods of describing the fluid flows. 3+3=6
  - (b) Determine the acceleration at the point (2, 1, 3) at t = 0.5 sec, if u = yz + t; v = xz - t and w = xy.
  - (c) Show that  $(x^2 / a^2) \tan^2 t + (y^2 / b^2) \cot^2 t = 1$  is a possible form of boundary surface of a liquid. 5
- 2. (a) Test whether the motion specified by

$$\vec{q} = \frac{k^2(x\hat{j} - y\hat{i})}{x^2 + y^2}, (\kappa = \text{constant}) \text{ is a possible motion for an}$$

incompressible fluid. If so, determine the equation of streamlines. Also, test whether the motion is of the potential kind and if so, determine the velocity potential. 2+2+1+2=7

(b) Derive Euler's equation of motion for inviscid fluid in vector form and hence generate Lamb's hydrodynamical equations. 5+2=7

## UNIT-II

- 3. (a) Verify that the curves of constant velocity potential and constant stream functions cut orthogonally at their points of intersections.
  - (b) Find the image of a doublet with regard to a circle.

(c) Two sources, each of strength *m* are placed at the points (-a, 0), (a, 0) and a sink of strength 2m at the origin. Show that the strength are curves  $(x^2 + y^2)^2 = a^2(x^2 - y^2 + \lambda xy)$ , where  $\lambda$  is a variable parameter. Also, show that the fluid speed at any point is  $2ma^2$ 

 $\frac{2ma^2}{r_1r_2r_3}$  where  $r_1, r_2$  and  $r_3$  are the distances of the points from the sources and the sink. 4+2=6

- 4. (a) If the expression for stream function is described by  $\psi = x^3 3xy^2$  determine whether flow is rotational or irrotational. If the flow is irrotational then indicate the correct value of the velocity potential.
  - (b) If  $\phi = 3xy$ , find x and y components of velocity at (1, 3) and (3, 3). Determine the discharge passing between streamlines passing through these points. 1+2=3

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(c) State and prove the Milne-Thomson circle theorem.

# UNIT-III

- 5. (a) Derive complex potential due to a rectilinear vortex filament. 7 (b) Prove that for the complex potential  $\tan^{-1} z$  the streamlines and equipotentials are circles. Find the velocity at any point and examine the singularities at  $z = \pm i$ , 5+2=7
- 6. (a) Determine the stream function when the strength of the vortex filaments are equal.
  - (b) An infinite single row of equidistant rectilinear parallel vortices of same strength *k* is at a distance *a* apart. Obtain complex potential, velocity components and streamlines. 4+2+2=8

# UNIT-IV

7. (a) Discuss plane Poiseuille flow.

(b) The stress tensor at a point is given with respect to the axes x, y, z

by the values 
$$\sigma_{ij} = \begin{pmatrix} 6 & 2 & 0 \\ 2 & 3 & 0 \\ 0 & 0 & 4 \end{pmatrix}$$
.

Determine the principal stress values and principal stress directions represented by the axes x', y' and z'. 7

- 8. (a) Discuss Hagen-Poiseuille flow. 7 3
  - (b) Explain the significance of Reynolds number.

(c) Velocity field at a point is given by 1+2y-3z, 4-2x+5z, 6+3x-5y. Show that it represents a rigid body motion. 4

## UNIT-V

- 9. (a) Discuss the elementary analysis of normal shock wave in brief. 7 (b) Derive the equation of motion of a gas in an isentropic flow process. 7
- 10. (a) Define nozzle. Discuss the flow of a gas through a nozzle. 1+6=7
  - (b) In a duct in which air is flowing, a normal shock wave occurs at a Mach number 1.5. The static pressure and temperature upstream of the shock are 170 kN/m<sup>2</sup> and 23°C respectively. Assuming  $\gamma = 1.5$ , calculate pressure, temperature and Mach number downstream of the shock. 3+2+2=7